



U.S. Department of  
Transportation  
Federal Aviation  
Administration

## MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved  
OMB No. 2120-0020  
2/28/2011

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

<b>1. Aircraft</b>	Nationality and Registration Mark <b>N76372</b>	Serial No. <b>10793</b>	
	Make <b>Cessna</b>	Model <b>140</b>	Series
<b>2. Owner</b>	Name (As shown on registration certificate) <b>Logan G Denelsky Robert L Bedard</b>	Address (As shown on registration certificate) <b>Address 179 Johnson Rd City Wadsworth State Ohio Zip 44281-9094 Country USA</b>	

**3. For FAA Use Only**

The technical data identified herein has been found to comply with the applicable airworthiness requirements and is hereby approved for use only on the above described aircraft, subject to conformity inspection by a person authorized in FAR Section 43.7

03/11/2019  
DATE   
FAA INSPECTOR, GL25

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial Number
<input type="checkbox"/>	<input checked="" type="checkbox"/>	AIRFRAME	—————	(As described in Item 1 above)	—————
<input type="checkbox"/>	<input type="checkbox"/>	POWERPLANT			
<input type="checkbox"/>	<input type="checkbox"/>	PROPELLER			
<input type="checkbox"/>	<input type="checkbox"/>	APPLIANCE	Type		
			Manufacturer		

### 6. Conformity Statement

A. Agency's Name and Address		B. Kind of Agency	
Name <b>Charles M. Widowski, Jr.</b>		<input checked="" type="checkbox"/> U.S. Certificated Mechanic	<input type="checkbox"/> Manufacturer
Address <b>2811 Norris Avenue</b>		<input type="checkbox"/> Foreign Certificated Mechanic	C. Certificate No.
City <b>Parma</b>	State <b>Ohio</b>	<input type="checkbox"/> Certificated Repair Station	<b>A&amp;P 3352309</b>
Zip <b>44134</b>	Country <b>USA</b>	<input type="checkbox"/> Certificated Maintenance Organization	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B <input type="checkbox"/>	Signature/Date of Authorized Individual
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### 7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is  APPROVED  REJECTED

<b>BY</b>	<input type="checkbox"/> FAA Fit Standards Inspector	<input type="checkbox"/> Manufacturer	<input type="checkbox"/> Maintenance Organization	<input type="checkbox"/> Person Approved by Canadian Department of Transport
	<input type="checkbox"/> FAA Designee	<input type="checkbox"/> Repair Station	<input type="checkbox"/> Inspection Authorization	Other (Specify)

Certificate or Designation No.	Signature/Date of Authorized Individual
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### NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

#### 8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N76372

Nationality and Registration Mark

Date

Replaced original hollow axles Cessna p/n: 0441140, item 13 of Figure 26, page 50 of the Cessna 120, 140 Parts Catalog with new Grove solid axle p/n: 5016 designed to fit Cleveland 6.00 - 6 wheels. Installed with new hardware and checked alignment and found to be within limits using appropriate shims. Tightened bolts to standard torque referencing AC 43.13-1B Table 7-1 on page 7-9. See attached Grove Drawing No. 5016 dated 7/24/2010 for axle bolt hole dimensions, previously approved Form 337 and a copy of a letter from Ronald K. Rathgeber, Manager of the Wichita Aircraft Certification Office dated June 27, 2001. See attached ICA checklist. There will be no weight and balance changes.

END

Additional Sheets Are Attached